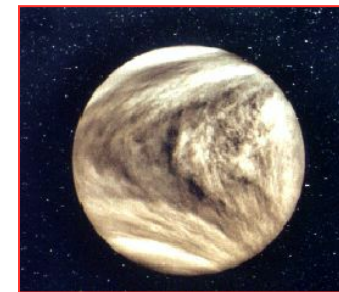


Homework, Section 1

- A sample return Probe is being sent on a 1-year mission from Earth to Mars Via Venus Using “aero-gravity” assist (*both both gravity And aerodynamics at Venus used to turn the corner to Mars*)



- The aero-assist maneuver at Venus is performed at An altitude of *110 km* above the Surface at a peak Atmospheric velocity of *13.09 km/sec*

- *At 110 km altitude, the ambient temperature is 147.63°K (-125.52 deg. C)*

- Assume that the Venusian atmospheric Composition by volume is {97% CO₂, 3% N₂}

- Calculate the Probe Mach number at the Venus aero-assist interface

**Assume that these
properties are constant
between -100 C and
-125.52 C)**

c_p At -100 °C

CO₂ ~ 0.845kJ/kg °K

N₂ ~ 0.995kJ/kg °K

Note Units!

Homework, Section 1 (cont'd)

- Show that for a reversible process

$$s_2 - s_1 = c_v \ln \left[\frac{T_2}{T_1} \right] + R_g \ln \left[\frac{\rho_1}{\rho_2} \right]$$

- and that for a reversible, adiabatic process

$$\left[\frac{T_2}{T_1} \right] = \left[\frac{\rho_2}{\rho_1} \right]^{\gamma-1}$$

Homework, Section 1 (cont'd)

- Show that for an ideal gas the following Useful relationships *hold*

$$c_p = \frac{\gamma}{\gamma - 1} R_g$$

$$c_v = \frac{1}{\gamma - 1} R_g$$